

While everybody talks about "clean design," the equivalent flat plate area of this one is small indeed. Its lines reveal its inheritance from the author's Formula 1 racing models.

# the SOLUTION

■ Hal deBolt



A new approach to AMA-FAI pattern, this machine weighs only 5.4 pounds, yet has fine performance with a 40 engine without a pipe. Borrowing from race plane experience, the front end incorporates various features that boost efficiency.

I HOPE YOU will not be shocked to see a pattern design offered by Hal deBolt. I have not participated actively in Pattern for some time, having been deeply engrossed in racing. But I have not lost my interest in an event which provided me so many great years.

There has not been much change since my last time around, except for the aircraft. Today's pattern aircraft are distinctly different from those of just a few years ago. You normally would say that this is just progress. However, in watching events from a designer's standpoint I have had reasons to wonder. I hear rumblings from the pattern troops that many are not satisfied with what the pattern airplane has developed into. In the minds of many, the speed, weight, initial cost, and operational costs of the latest designs are undesirable. The question has to be, is there another way?

There is another way, if you are willing to depart from the "norm." Wondering why things are being done as they are, plus the introduction of some new equipment, encourages me to suggest the "Solution." It would be great if I could tell you that the Solution is a world champion, and the greatest pattern design ever, but I cannot. Although the Solution inherits the characteristics of past "champions," it really provides ideas and methods you can use to obtain desired results. The Solution is a thoroughbred pattern craft in its own right.

You could do worse than give it a try!

So what is so different? To understand the difference you must be familiar with the progress of pattern designs during the past few years, and know whether what has been done had a positive or negative influence. Reasons for the changes make little difference. What is important is the end result, and how it affects the aerodynamics of a pattern aircraft. The first change came in wing area and size. Wing size was reduced by about 20 percent, when from an aerodynamic standpoint, an increase in size would have added efficiency. Fuselages became more bulky, engines were located more and more in the open. Then came the muffler rule with that big gob of weight hanging out in the breeze. All this, of course, did little more than add drag. Next was the decision to depart from the proven wood structure for fiberglass and foam, which did nothing aerodynamically but add weight and in-

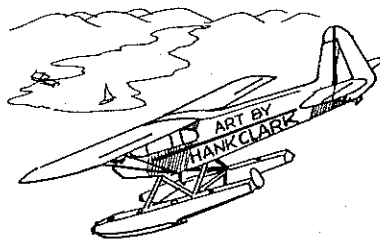
crease wing loadings.

While doing all this, the need remained to maintain, or even improve, performance. There is only one way to counteract added weight and drag, and that is to increase power and speed. Unfortunately, when we counteract with power, the percentage of power increase has to be much greater than the percentage of increase in the drag and weight you are attempting to overcome. Additional power requires a larger and/or heavier powerplant. This adds more weight and increases fuel consumption. In short, the whole thing becomes a vicious cycle: reduce wing area, add weight with materials, increase drag, and then counteract with more power—which adds more weight, resulting in the need for even more power. There is no end to such a cycle and it appears that today's normal pattern design is the result of such a cycle.

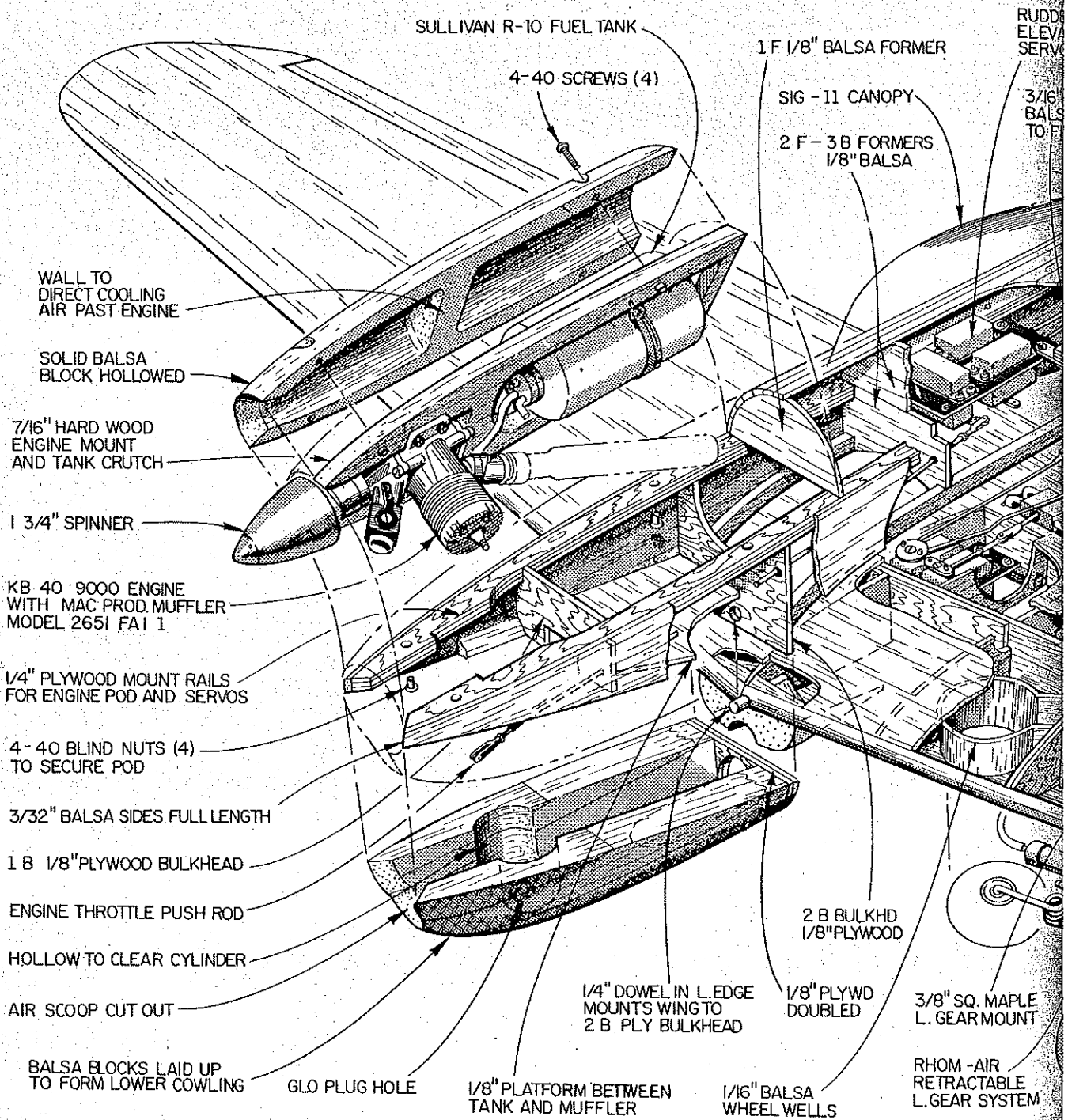
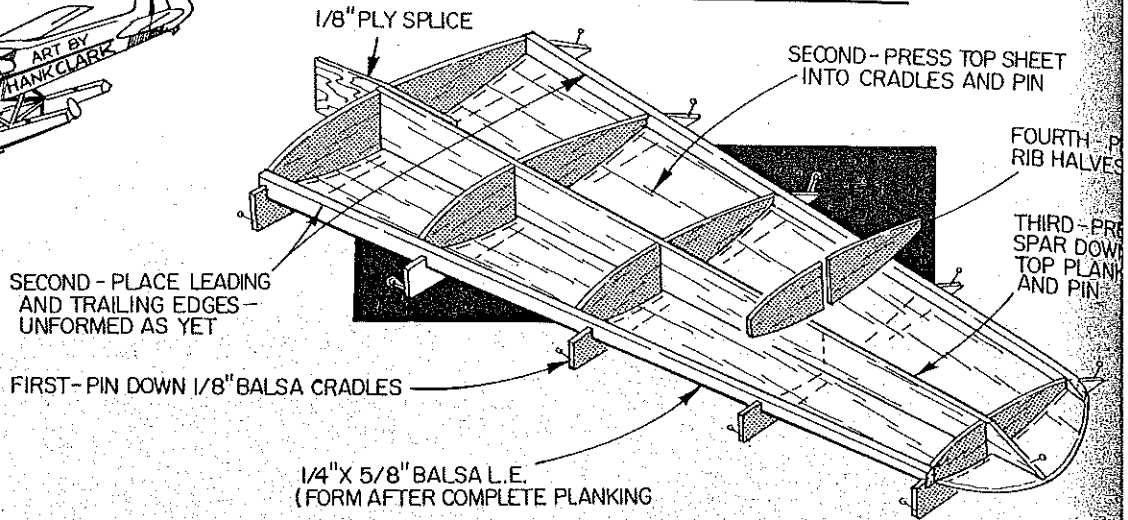
The serious designer remembers one

### WEIGHT BREAKDOWN

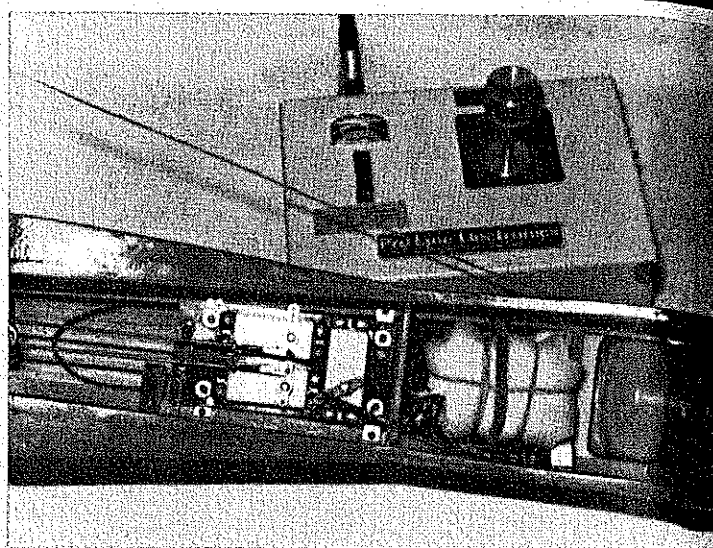
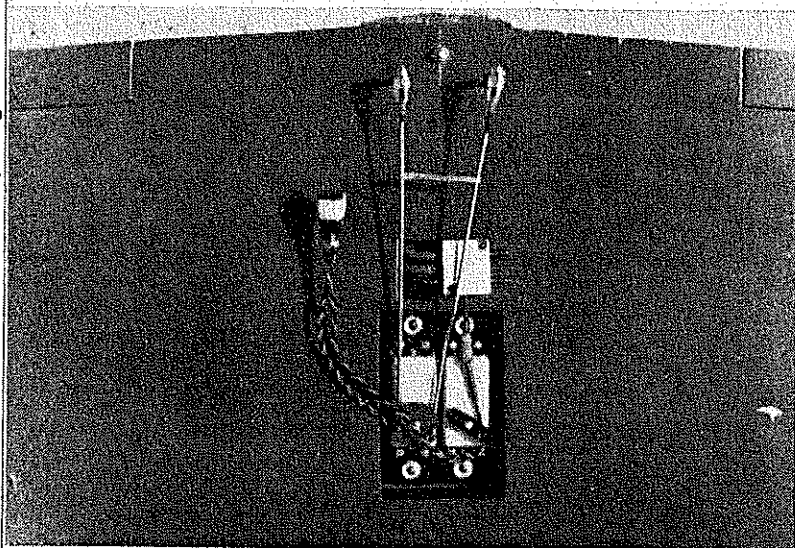
Finished Structure		Covered & Painted	With Equipment
Wing	14 oz.	21 oz.	33 oz.
Tail	3 oz.	23 oz.	53 oz.
Fuselage	15 oz.		
Totals	32 oz.	44 oz.	86 oz. (5.4 lbs.)
Total aircraft weight—44 oz.			Total equipment weight—42 oz.



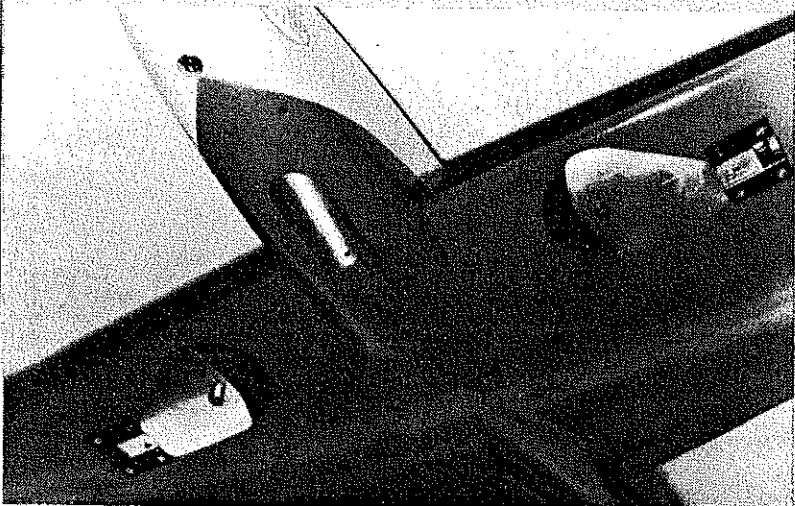
**JIG FOR WING ASSEMBLY**







Left: Aileron and retract servos mounted side by side—strip ailerons well outboard of root. Right: Sport fliers note: this is the way to install radio.

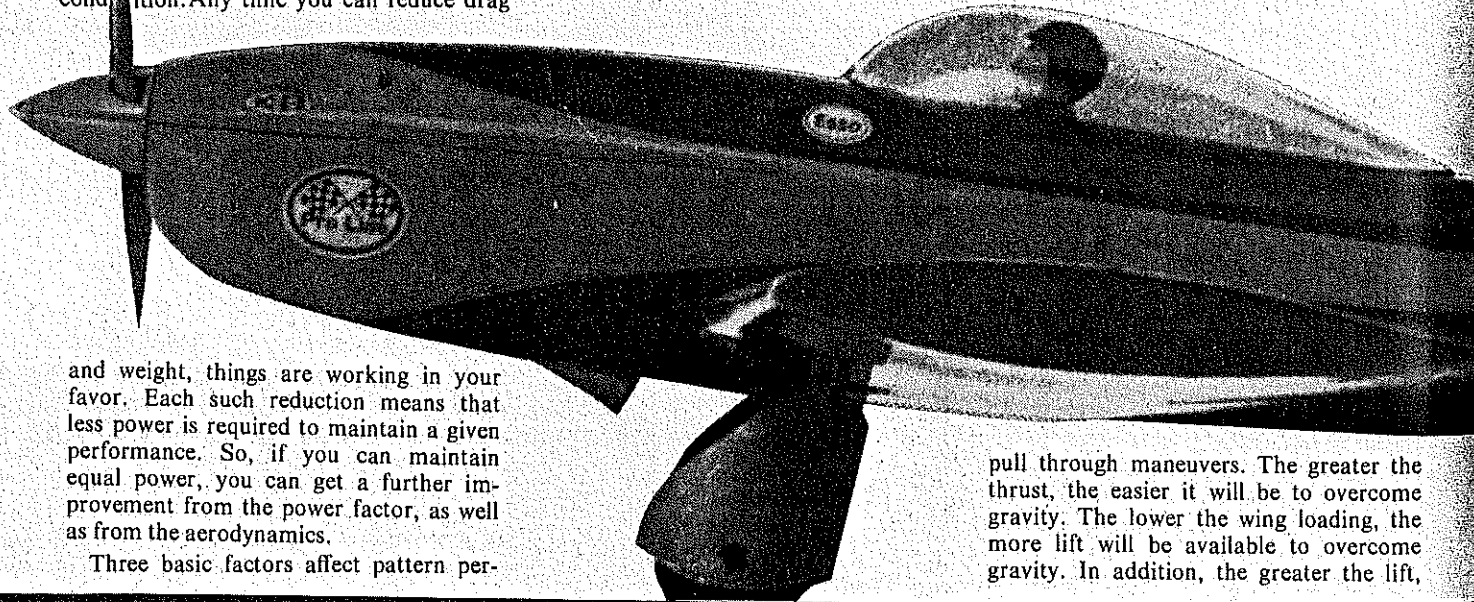


Left: Exiting neatly, forward of the wing, the Mac's Products Special muffler for the K&B engine, is one key to the front-end design. Right: Annular cooling opening around the spinner aids in cooling the crankcase, too often ignored. Cylinder is shrouded for maximum cooling.

axiom. Adding weight and drag compounds a negative problem. Reducing weight and drag compounds a positive condition. Any time you can reduce drag

How's this for a nifty profile? Ultra light and streamlined, the Solution can climb out of sight vertically while rolling, on its 40 engine turning an 11-7—if anyone wants to!

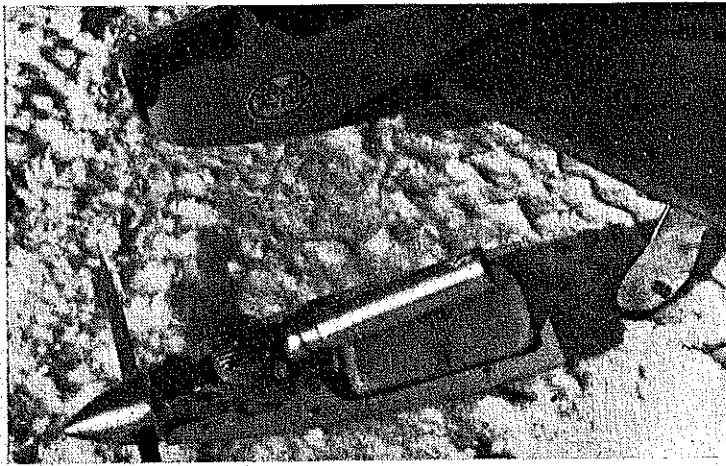
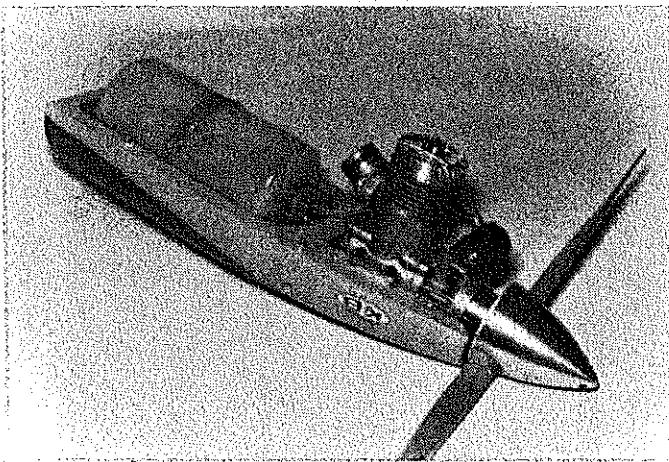
formance: drag, thrust, and wing loading. The less the drag, the greater will be the ease of flight, the more power available to



and weight, things are working in your favor. Each such reduction means that less power is required to maintain a given performance. So, if you can maintain equal power, you can get a further improvement from the power factor, as well as from the aerodynamics.

Three basic factors affect pattern per-

pull through maneuvers. The greater the thrust, the easier it will be to overcome gravity. The lower the wing loading, the more lift will be available to overcome gravity. In addition, the greater the lift,



Left: The combination of the engine and its muffler yield a lightweight power unit that provides power usually associated with engines nearly double the size and weight. Right: Utility and efficiency is gained by removable pylon-type pod providing solid mounting for engine and tank.

the *less thrust* is required. All these factors are interrelated. The object of good design is to use them in the best compromise, for compromise you must!

What made the objectives so easy to accomplish was the recent introduction of the K&B 40 model 9000 engine, plus the Mac's Products special muffler to use with it. Normally, I am not the one to design around a particular product and, for the Solution, doing so probably was not absolutely necessary. There are other engines which provide ample power. However, the model 9000 amplifies the advantages of the design. This engine really does have the power of a normal 60 with about half the weight, at a considerably smaller size. Less engine weight and size means less structural weight and lower drag. Lower fuel consumption helps. This 40 will turn a 60-size propeller at, or higher than, the rpm of the *normal* 60.

So, what is so hot about the Solution? Its structure and aerodynamics are different. If there is any one outstanding thing you learn from pylon racing it is how to build light but strong and durable airframes. Much of that I learned from racers has been applied to this design. One might question the use of a power pod. The utility is obvious, but the fact that it is an excellent power and vibration absorber is not. It also provides a neat way to enclose

necessities needed with other types of mounts disappear with a pod, so any extra effort is worthwhile.

The wing is the heart of the aircraft. The greater the efficiency of the wing, the less the remainder of the aircraft must contribute. To a great extent, the wing controls the flying speed because of the drag it produces. Drag is affected by the size of the wing, and the airfoil used. A very efficient low-drag airfoil is used, and wing size is greater to control speed. With the larger wing, lift is greater. The center airfoil is a NACA 65015, about the finest foil available for our use. The thickness is at the top of the efficiency range; anything thicker would do nothing but add drag. An airfoil progression also is used, changing from the 65015 at the root to a 65012 at the tip. The 65012 is more stable and has less drag. We wish the tips to be stable, of course, and with less tip drag, the center of drag is moved in closer to the center of the aircraft. We wish to concentrate the drag and weight as close as possible to the center of the craft for greatest stability. Efficiency would be greater with a higher aspect ratio but a compromise is necessary when maneuverability is required.

Dihedral is desirable but is often discarded in recent designs. It is obtained here in a sneaky way. Because of the wing taper, a major portion of the dihedral is obtained by putting all of the thickness taper into the bottom of the wing. In upright flight there is positive dihedral action,

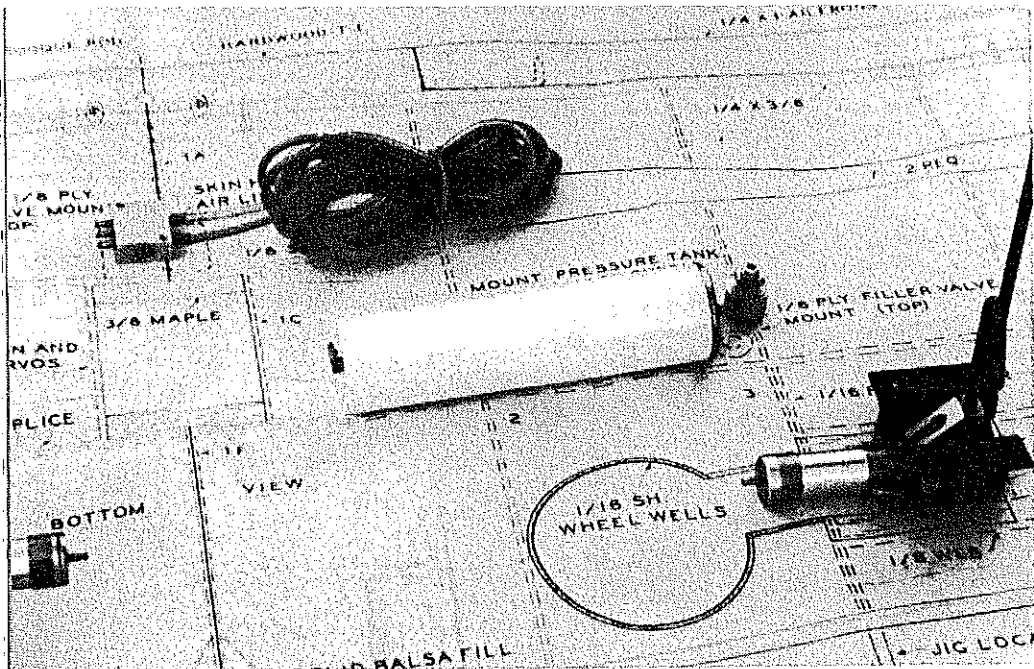
needed to maintain attitude or for righting action. You will find very few real aircraft that do not use dihedral.

Both tail surfaces use airfoils similar to that used in the wing. The tail creates lift as necessary to control the wing. If lift is needed, you get more of it with *less drag* from a good airfoil, than from a flat plate. The aircraft is more efficient with tail airfoils. With greater efficiency we can use less tail area. The shape of the vertical tail is deliberate. Note that the center of the rudder area is near the thrust line and that the hinge line is slanted to the thrust line. Such an arrangement removes any rolling action when rudder is applied in maneuvers.

The horizontal tail is minimum in size because of a good airfoil. With such a tail the elevator movement is minimized. As the elevator is moved, the airfoil changes from full symmetrical to a form of under-camber, much like a lifting flap. The increase in the amount of the lift thus created is drastic. Once again drag is minimized. As to the fuselage, you have greater efficiency with less bulk and less

the engine and muffler for drag reduction. Another plus is the pressure cowl. It aids engine cooling and further reducing drag. Building time is only slightly more. Many

yet when inverted, there is little dihedral to detract from stability. Since dihedral provides automatic stability, no control is



When Hank Clark was making the cutaway drawing he snapped this picture of the Rhom Air retract system parts laid out on the author's plans. The two-wheel gear eliminates weight of nose unit.

drag. In designing his very successful Formula I racers, George Owl concentrates the bulk of the fuselage at the nose and tapers straight back. This is done in the Solution.

You cannot do much better than a fully cowled engine. However, the manner in which it is cowled can yield improvement. A pressure cowl is used to reduce drag and improve engine cooling. A simple "helmet" cowl will not do this. Mac's muffler complements this cowl design, since it is fully enclosed.

Something really new in cowl air intake design is used. The annular air intake design has only recently proven to have outstanding advantages in Formula I aircraft. Any intake system disturbs the air flow around it. With this system, the air is taken in around the spinner where the air is disturbed no matter what is done. As a result the nose of the cowl is smooth, re-

ducing drag. The intake area is larger, allowing more air to enter. The advantage is that the entire engine receives cooling air, not just the cylinder as with "slot" intakes. For proper cooling an air flow must be created past the entire engine. This does not happen when the engine is mounted in a "pocket" as in most pattern designs. The cylinder is out in the breeze but very little air moves past the crankcase.

The fuselage includes wing fillets. They are always a pain, so the minimum you have the less the pain. The fillets occur only in the necessary area. Fillets are a distinct asset. We use abnormally thick airfoils in our pattern aircraft. Considering scale effect and all, we fly these foils at speeds above the desirable range. So it does not take much to make the air turbulate off the trailing edge. This turbulation occurs most easily where the wing joins the fuselage. This turbulent air flows

back along the fuselage and is further agitated. Under extreme conditions, the turbulent air can affect the vertical tail. Without "solid" airflow the tail can oscillate. We call this "fish tailing." Wing fillets help provide the tail with the solid airflow it needs.

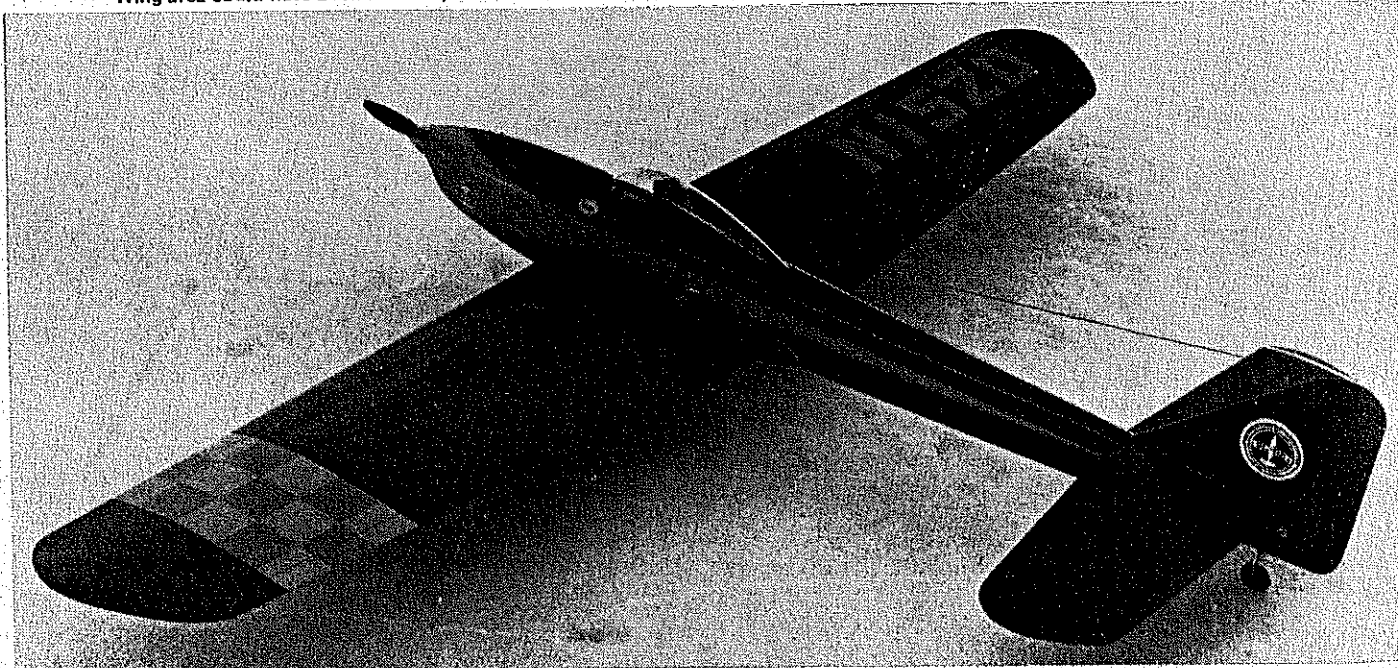
The force arrangement of any aircraft is important in that it determines whether the design does its work efficiently or not. The Solution's force arrangement is well proven. A considerable amount of work went into its development. Over the years it paid off with a "bit better" flying model than any other arrangement I tried. The symmetrical airfoil develops no lift at a zero angle of attack. It must be at a positive angle to gravity to create the required lift. We fly upright most of the time, a smaller portion of the time inverted. If we set the airfoil at zero we must force the entire aircraft into a positive angle of attack when either upright or inverted. This creates drag. If we set the airfoil at a positive incidence angle great enough for the wing's lift to equal the weight of the aircraft, it will remain level when upright, keeping drag to a minimum. One degree positive is enough to create this condition. When inverted the angle of attack will have to change and drag will be created—it is a case of choosing the lesser of two evils.

With the wing's angle of incidence fixed, an angle for the stabilizer has to be found which will create tail lift proportional to the wing's lift. With the proper arrangement the model will tend to fly straight no matter what the speed change may be. With the tail lift created by size and airfoil already determined, an angle of 1½ degrees positive will create the desired relationship.

With a low-wing aircraft such as this, the center of resistance will fall above the

*Continued on page 85*

Wing area could have been reduced, but was not. Combination of generous area and excellent airfoil gives superb performance.



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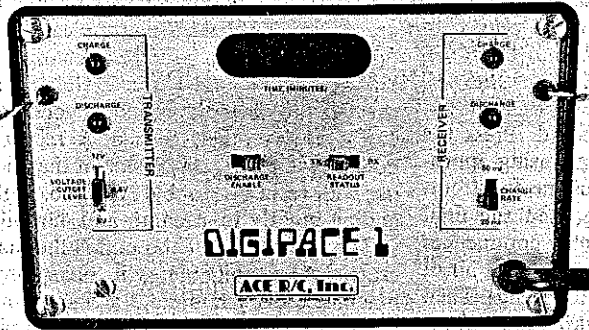
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interference can easily get in at that point. This is particularly true if the shielded wire making the connection is broken, or if the connections are dirty. You have to check out these possibilities before making a decision. Ultimately, the simple solution is to keep your transmitter antenna collapsed as much as possible, and to seek out the locations and alignments of the antenna that produce the least interference.

Several people have requested information on Art Cervenka's castoring nose wheel for the Ranger 42. There are many ways to get the effect. The one shown in Fig. 3 works. All you need are some pieces of 3/32 I.D. brass tubing, a piece of 3/32 O.D. music wire, some 20-gauge copper wire, solder and a rubber band.

Keep those letters coming folks; it makes me feel wanted!

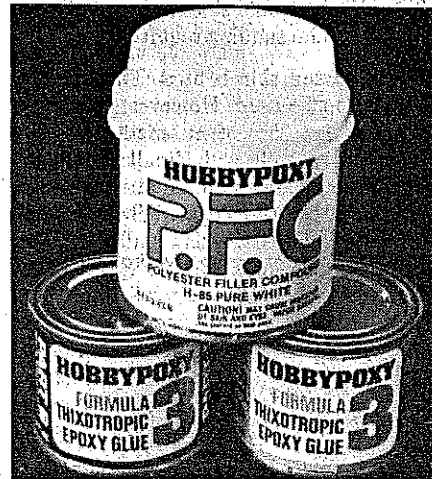
George Myers, 70 Froelich Farm Rd., Hicksville, NY 11801.

## Solution/deBolt

continued from page 18

wing, considering flat plate drag. Drag also increases as the square of the speed. As speed increases, the drag of a wing rises more quickly than for the rest of the craft. As far as maintaining straight flight over

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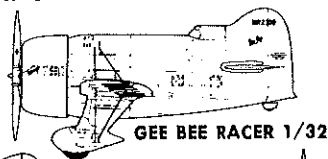
a wide speed range is concerned, we need something to compensate for the shift in the center of drag caused by the wing. This is accomplished by placing the line of thrust above the center of resistance. Thrust decreases proportionally to speed; this action is used to counteract the wing drag, which also is proportional to speed.

To obtain the optimum performance from an aerobatic model you need one which will require the least amount of control. This quality is called "pointing," or neutral stability. Once pointed, the craft stays exactly on that course until control action changes its heading. When completing a maneuver, neutralizing the controls turns the control of the flight over to the aerodynamic controls designed into the model. The force arrangement—more than pilot ability—contributes the most to this action.

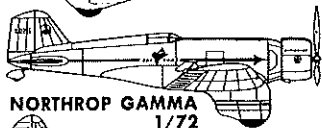
We could continue to describe the little design details which add to the overall quality of the Solution. Such things as wing taper using a bit of sweepback certainly are important, as are all of the details. However, you will recognize these things. Pattern design is not new. It is important to have the proper basics, then add details. The Solution does this because of the abnormal attention paid to drag reduction, the unusual amount of lift available from the efficient wing, light wing loading and, of course, the ample thrust provided by the powerplant.

Windy weather performance is bound to

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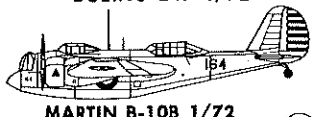
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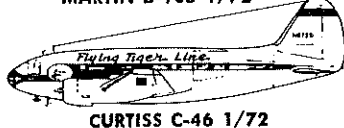
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be a question. A model will have good rough air performance if it penetrates well and does not change heading as it drifts with the wind when flying cross wind. Low drag and ample thrust provide good penetration. A properly located center of lateral area will help maintain headings in a cross wind.

### Construction

Since I wanted a low wing loading and optimum performance, no foam or fiber glass is used. (Templates are provided for a foam wing.) A foam wing will add at least a 1/2 lb. With foam the C.G. of each panel will be moved farther out on the wing with a loss in wing stability. A fiberglass fuselage could be produced, but the nose length would probably have to be increased for balance.

**Wing:** Structural strength is enhanced by the use of stressed skin. Practically all of the strength comes from the outer skin. In this case, the skin *must be* a laminate. Although the wing has wood covering, it is not satisfactory to just fill the wood by the resin method, or cover it with the stretchable plastics. A fabric needs to be tightly bonded to the sheet covering. I used Silron fabric attached with common dope. It goes on easily and once in place, requires little sanding and filling. An alternate is 3/4-oz fiberglass cloth applied with resin—more work and a bit heavier. With the stressed skin principle much of the in-

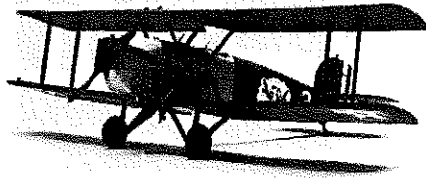
ternal structure is eliminated, adding to simplicity. The landing gear mounts, etc., are used to do the work additional structure once performed.

The wing assembly method is a recent development which assures an absolutely true wing, making the assembly simply a job of "sticking" parts together. A dihedral board is a must. The jig is set up so that the bottom of the wing faces up; the angle of the board will match the dihedral of the top of the wing. Wing assembly "saddle jigs" are fabricated. Using the wing plan for dimensions, the locations of these jigs are determined on the board, then they are secured in place. Sheeting for the top skin is glued up and roughly sized. Press the sheeting into the saddle jigs, using appropriate ribs. This determines the exact outline of the wing sheeting, and the final sizing can be made. Assembly commences with the spar, followed by the ribs, all of which form the sheeting into the jigs. Then it is simply a matter of sticking the various parts in place, and finishing the main structure by planking the bottom with 3-in. sheeting.

The wing tips are different. No blocks are used, saving several ounces in a critical area. No carving or shaping is needed. When the wing is out of the jig, the top sheeting will extend past the end ribs. This is cut to the tip shape shown. Then with a large flat sanding block, the bottom is trued up to accept the tip plate. This is done by sanding fore and aft, so that the tip plate will mate neatly to the end rib on the bottom and the edge of the tip sheeting on top. The tip plate should form a smooth surface with no sanding required.

**Tail:** Stressed skin is used. The same jig method can be used. However, you might want to use the other method shown which works well with small structures. This is the centerline method. Begin with leading and trailing edges. When they are

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placed on the building board, their width is great enough that no portion of the structure will touch the board. Centerlines are scribed on all the ribs as well as the leading and trailing edges. With the edges in place, insert the ribs between them, obtaining alignment with the centerlines. Once they are fastened, add the sheeting and, when ready, flip over the structure and add the sheeting on the other side. Shaping completes the job.

**Fuselage:** Except in the nose cowl and pod area, the fuselage is the typical "box" faired on the top, and for a change, the bottom. We "planked" the top fairing, and sheeted the bottom. Planking goes on as easily as any other method. A recent innovation is to coat the inside of the planking with resin or Hobby Pox No. 2 glue, adding greatly to strength and durability.

The foundation of the pod is the maple crutch. As formidable as it looks, simple machine tools can make it a cinch—and it costs only pennies! Scrounge a small piece of 5/4 (1 1/8") maple from the school shop or lumberyard scrap pile and you have the making of two crutches. The wood is easily shaped with a scroll, jig, or band saw. It can even be done with a drill press. Drill a zillion holes around the outline and break away the scrap. With a straight sided router bit, the true outline can be ground to shape. If you shaped the full 5/4 maple, the crutch can be run through a table saw and presto, you have two!

The fuselage mount for the crutch is 1/4" plywood, a simple shaping job. In locating the shear pins and hold-down screws, first position them in the crutch. Then, with the crutch clamped to the fuselage mounts, use the crutch holes as a drill guide.

Cowlings often are a chore. If you enjoy carving and shaping balsa blocks, the described method of construction avoids the usual agony. Assemble the cowl in layers; engines vary in shape from the crankcase up, as does the clearance required. Mount the engine, hook up the throttle linkage, and chop a hole into a piece of 1" balsa. Slip the balsa over the cylinder, then enlarge the hole by eye until necessary clearances are attained. Cement this first layer to the fuselage; precision fits are not required. The next layer covers the cylinder fins and head. Make a 1 1/2" diam. hole in another piece of 1" stock, and slip it over the cylinder. The fit to the cylinder should be close, and clearance for the carburetor and muffler will need to be provided. The final layer will simply cap off the cylinder head, and fair the whole into the bottom of the fuselage. As to the outside, you can rough shape the layers as you build them up, otherwise you will have a big glob of balsa to attack with your carving knife. The sizing of the outside is not difficult because you can hack off huge chunks during the rough shaping. It is important to bring the final shape down to a minimum while maintaining smoothly flowing curves.

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favorite ways. The one I used (Silron and dope) works extremely well, requires little effort, and provides excellent durability. Cover forward section of the fuselage and the pod with ¼-oz. fiberglass cloth and finishing resin. This is durable, and 100% fuelproof. Cover the aft section of the fuselage and tail with Silkspan paper and dope. Finish with one heavy coat of Super Poxo primer, followed by colored Hobby Poxo as desired. The final step is a coat of clear Hobby Poxo to seal everything in place. If you wish a super finish, sanding out the clear and adding more coats will give you the optimum.

We hope the Solution will add to your flying pleasure. We know it will save you money.

### RC Aerobatics/VanPutte

*continued from page 21*

"Since adopting this tailwheel shoe system I have not lost a single empennage due to snagged tailwheels. If the tailwheel runs into a gopher hole, the rubberband breaks, the tailwheel shoe comes free of the fuselage, the plastic clevis pops open, and the plane lands safely, ready to be flown again after assembly with a new rubberband."

A short time ago I received a note from the editor which contained the following information: "We have had fantastic success with the Simitar; it is our number one plan. As of two months ago, Bill Evans had sold

more than 3,000 wing cores. This indicates a state of fusion that goes beyond the magazine. I saw this happen only once before when *Model Airplane News* sold more than 8,000 Smog Hog plans. What makes this new information interesting is the variety of types that Bill had derived from the original design."

There's no doubt that a different looking airplane always draws interest at the flying field. If it flies well, there is bound to be a lot of copies on the building board shortly afterward. That's what had happened to Bill Evan's Simitar. The Simitar XV first

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Seen at Toledo: Four-bladed, rigid rotor, fully aerobatic helicopter. American RC Helicopters, Inc., 23811 Via Fabricante, Mission Viejo, CA 92675.

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appeared as a construction article in the December '76 issue of *Model Aviation*. Nationwide interest and response to this definitely different craft gives every indication that the Simitar design will be given credit for making the flying wing a practical reality within the grasp of all RC pilots.

Bill has designed and built no less than 30 flying wings. Among these have been both gliders and powered airplanes. Gliders: the Saracen (*RCM*, April 1976), a 72" glider; the Little Saracen, a 48" glider; and the Super Saracen, a 120" glider. Power: ½A Simitar (*RCM*, December 1976) and Simitar XV (*MA*, December 1976). The Simitar XV now has been improved upon to include motor control and steerable tricycle landing gear, an epoxy fiberglass fuselage (with molded-in canopy, air intakes, firewall and removable engine cowl). With a Simitar XV and a K&B .40, you have the Simitar 540.

But that's not the end; Bill has now developed a series of Simitar Twins. The first, pictured above, is powered by two Cox Medallion .09's. This is two channel, only using elevons, with no engine control. After one engine quits this 3½-pounder maintains altitude on *one* .09. There is also a .049 Twin as well as a .19 Twin now flying. A .40 Twin is soon to come.

What next? Well, Bill says that his bottom line will be to complete the negative stagger twin-engine Simitar Bi Wing. Guess we'll have to wait to see that!

Next month's column will include the